

## ENGINE TYPE 912 UL I 80 hp

## DESCRIPTION

- 4-cylinder
- 4-stroke liquid/air-cooled engine with opposed cylinders
- dry sump forced lubrication with separate oil tank, automatic adjustment by hydraulic valve tappet
- 2 carburetors
- mechanical fuel pump
- dual electronic ignition
- electric starter
- propeller speed reduction unit

912UL $\qquad$ Call for Price

## FACTS

This series was BRP's first Rotax engine dedicated for aircraft application only. The Rotax 912 series is well regarded for its reliability and efficiency and is primarily targeted as the entry level engine in the light aviation industry. The 80 hp version of the Rotax 912 series offers a time between overhauls of 2.000 hours and is available as non-certified (Rotax 912 UL ) as well as certified version according to FAR 33 (Rotax 912 F) and JAR 22 (Rotax 912 A).

| VERSION | PERFORMANCE |  |  | TORQUE |  |  | MAX <br> RPM |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\left.\left.912 \mathrm{UL}^{1}\right) / \mathrm{F}^{2} / \mathrm{A}^{3}\right)$ | $\begin{gathered} \text { kW } \\ 59.6 \end{gathered}$ | ft. lb. $80$ | $\begin{aligned} & 1 / \mathrm{min} \\ & 5800 \end{aligned}$ | $\begin{aligned} & \mathrm{Nm} \\ & 103 \end{aligned}$ | $\begin{aligned} & \text { ft. lb. } \\ & 75.9 \end{aligned}$ | $\begin{aligned} & 1 / \mathrm{min} \\ & 4800 \end{aligned}$ | $\begin{aligned} & 1 / \mathrm{min} \\ & 5800 \end{aligned}$ |

Limited for max. 5 min.

| BORE | STROKE |  | DISPLACEMENT |  | FUEL |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 79.5 mm 3.13 in 61 mm | 2.4 in | 1211.2 <br> $\mathrm{~cm}^{3}$ | 73.91 <br> cu. in | min. MON 83 RON 91* <br> min. AKI 87* |  |

* leaded, unleaded, AVGAS 100LL or Ethanol 10

1) UL = non certified $\left\lvert\, \begin{array}{lll}\text { 2) } F=\text { certified acc. to FAR } 33 & \mid & \text { 3) } A=\text { certified }\end{array}\right.$ acc. to JAR 22


ENGINE TORQUE


FUEL CONSUMPTION


## PERFORMANCE GRAPHS

| WEIGHT | kg |  |
| :--- | :---: | :---: |
| engine with propeller speed reduction unit $\mathbf{i}=2.27$ (opt. $\mathrm{i}=2.43$ ) | 55.4 | lb |
| overload clutch | 122.0 |  |
| exhaust system | 1.7 | 3.7 |
| external alternator | 4.0 | 8.8 |
| engine suspension frame | 3.0 | 6.6 |
| air guide hood | 2.0 | 4.4 |
| airbox | 0.8 | 1.8 |

