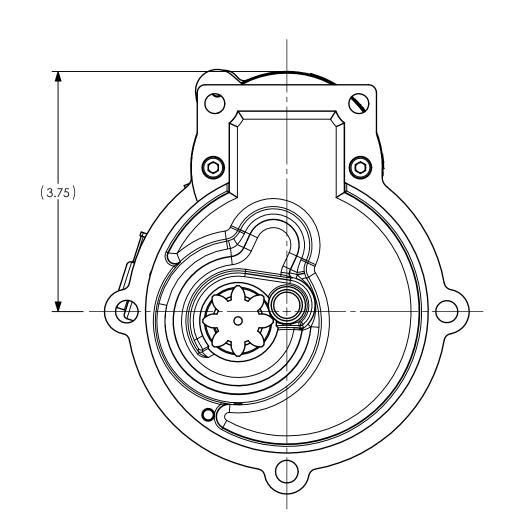
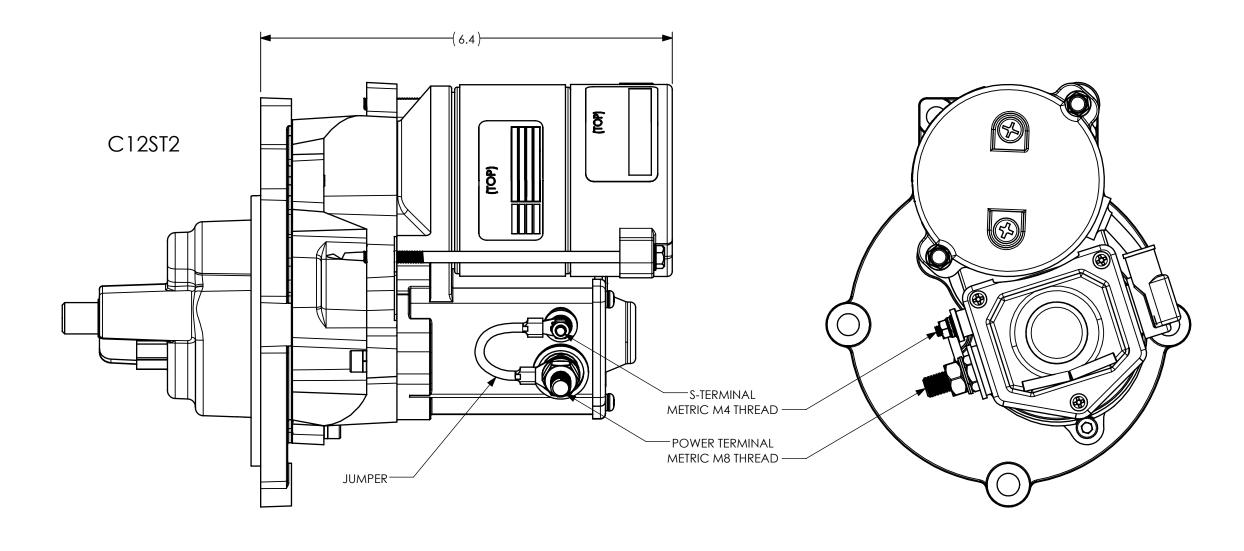
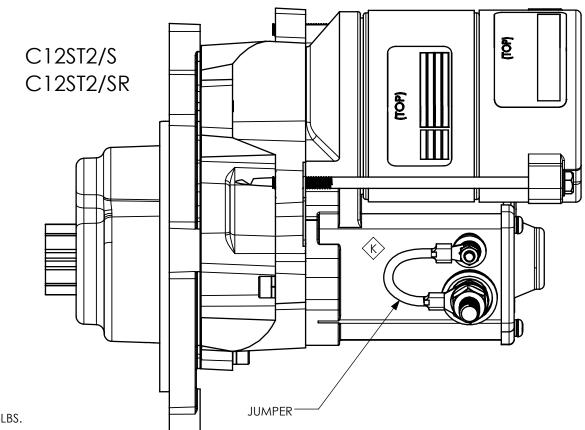
REVISIONS REV. DESCRIPTION ECO USER IS RESPONSIBLE FOR VERIFICATION 1) REDRAWN IN SOLIDWORKS; 2) FORMAT: HET WAS SKYTEC; 3) DESCRIPTION: CHANGED TO "ST2 INSTALLATION INSTRUCTIONS"; 4) REMOVED PMA DATA TABLE AND LABEL GRAPHIC OF CURRENT REVISION BEFORE USING 123514 THIS DOCUMENT. DOCUMENT CONSIDERED "REFERENCE ONLY" IF NOT 1) UPDATED VIEWS WITH JUMPER FOR ALL MODELS | EAB | 03/08/17 124308 THE CURRENT REVISION.







INSTALLATION INSTRUCTIONS:

- 1. DISCONNECT NEGATIVE (-) TERMINAL OF AIRCRAFT BATTERY.
- 2. DISCONNECT POWER CABLE FROM OLD STARTER AND REMOVE OLD STARTER FROM ENGINE.
- 3. INSTALL NEW STARTER WITH NEW GASKET USING EXISTING HARDWARE.
- 4. TIGHTEN HARDWARE TO TORQUES SHOWN IN TCM ENGINE OVERHAUL MANUAL (TYP. 275-325 IN-LBS.).
- 5A. FOR "KEY START" REPLACEMENT, ENSURE JUMPER WIRE IS INSTALLED BETWEEN STARTER POWER TERMINAL AND STARTER S-TERMINAL USING SUPPLIED HARDWARE. CHECK NUT TORQUE ON STARTER S-TERMINAL TO BE 25 IN-LBS.
- 5B. FOR "PULL START" REPLACEMENT USING STC SE10076SC & SA 10077SC, ENSURE JUMPER WIRE BETWEEN STARTER POWER TERMINAL AND STARTER S-TERMINAL IS REMOVED AND WIRE HARNESS IS INSTALLED. TORQUE S-TERMINAL NUT TO 25 IN-LBS.
- 6. INSTALL THE POWER CABLE ON THE POWER TERMINAL OF THE STARTER WITH A MS25171-2S OR SIMILAR TERMINAL NIPPLE AND LOCK WASHER & NUT SUPPLIED. TIGHTEN SUPPLIED METRIC M8 NUT TO 50 IN-LBS.
- 7. CHOCK AIRCRAFT AND CLEAR PROP BEFORE RECONNECTING NEGATIVE (-) TERMINAL OF AIRCRAFT BATTERY.
- 8. CONTINUED AIRWORTHINESS: THE STARTER SHOULD BE OVERHAULED BY SKYTEC AT 2700 HOURS OR WHEN A MAJOR OVERHAUL IS PERFORMED ON THE ENGINE ON WHICH IT IS INSTALLED. IN CASE OF STARTER MALFUNCTION OR FOR OVERHAUL, SEE WWW.SKYTECAIR.COM TO OBTAIN INSTRUCTIONS FOR RETURNING THE STARTER TO SKYTEC. NO OTHER MAINTENANCE IS REQUIRED OR RECOMMENDED.

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SPECIFICATION CLASSIFICATION							
CL ACCIFIC ATION	DIMENSION	NOTE NO.	SUR				
CLASSIFICATION	CONVENTION	CONVENTION	THIS CC ENGIN ON T THE IN				
CRITICAL	<xx.xx></xx.xx>	<#>					
MAJOR	[XX.XX]	[#]	OR DI				
MINOR	XX.XX	#	NOT E				
REFERENCE	(XX.XX)	(#)	TEC RE				

DIMENSIONS ARE IN INCHES AND APPLY AFTER HEAT TREAT AND PLATING				
.X = ±.015 .XX = ±.010 ANGLES ±1° .XXX = ±.005 BREAK ALL EDGES AND MACHINE ALL INSIDE CORNER FILLETS .015 MAX. 125' SURFACE FINISH V	1 10¤(1-1			
THIS DRAWING CONTAINS INFORMATION THAT IS CONFIDENTIAL AND PROPRIETARY TO HARTZELL ENGINE TECHNOLOGIES. THIS DRAWING IS FURNISHED ON THE UNDERSTANDING THAT THE DRAWING AND THE INFORMATION IT CONTAINS WILL NOT BE COPIED OR DISCLOSED TO OTHERS EXCEPT WITH THE WRITTEN CONSENT OF HARTZELL ENGINE TECHNOLOGIES, WILL NOT BE USED TO THE DETRIMENT OF HARTZELL ENGINE TECHNOLOGIES, AND WILL BE RETURNED UPON REQUEST BY HARTZELL ENGINE TECHNOLOGIES.	♥ = ∠//// SC			
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UNLESS OTHERWISE SPECIFIED

10/20/16	10/20/16	EAB	DRAWN	SEOMETRIC SYMBOLS	
1/ FAR ZELL \ 2900 Selma Highway	K. NORD		ENGINEER	PER ANSI Y14.5	
08 ENGINE TECHNOLOGIES Montgomery, AL 36108	308	124	APPRVD.	7 FLATNESS - STRAIGHTNESS	
	A	N/	FINISH	CYLINDRICITY	
A	'A	N/	WEIGHT	PERPENDICULARITY	
ST2 INSTALLATION INSTRUCTIONS	N/A		MATERIAL	POSITION CONCENTRICITY SYMMETRY ANGULARITY PARALLELISM	
CODE ID DRAWING NO. REV. 65PY1 70001 K		SH 1 OF 1	SIZE C	CALE 2:3	
ST2 INSTALLATION INSTRUCTIONS CODE ID DRAWING NO.	308 A 'A CODE ID	124 N/ N/A SH 1	APPRVD. FINISH WEIGHT MATERIAL	STRAIGHTNESS ROUNDNESS CYLINDRICITY PROFILE PERPENDICULARITY POSITION CONCENTRICITY SYMMETRY ANGULARITY PARALLELISM CIRCULAR RUNOUT	

BY DATE

EAB 10/20/16