

AVIATION MAINTENANCE TECHNICIAN CERTIFICATION SERIES

DIGITAL TECHNIQUES ELECTRONIC INSTRUMENT SYSTEMS

5





Update notices for this book will be available online at www.actechbooks.com/revisions.html
If you would like to be notified when changes occur, please join our mailing list at www.actechbooks.com

VERSION	EFFECTIVE DATE	DESCRIPTION OF REVISION(S)
001	2013.01	Module creation and release.
002	2014.07	Format updates and minor type corrections.
002.1	2021.05	Formatting Updates - no content changes.
002.2	2023.04	Inclusion of Measurement Standards for clarification, page iv. Minor appearance and format updates.
002.3	2023.06	Replaced duplicate Questions and Answers for Submodule 5, pages 5.9 and 5.10.
003	2024.05	Regulatory update for EASA 2023-989 compliance.

Module was reorganized based upon the EASA 2023-989 subject criteria.



TABLE OF CONTENTS

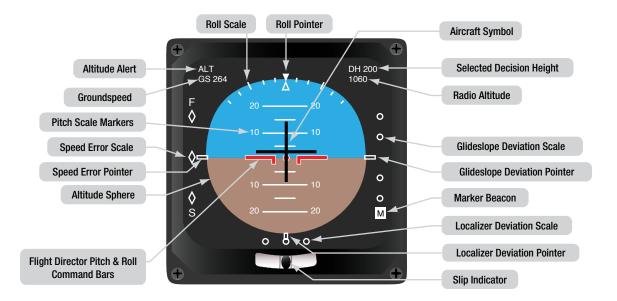
DIGITAL TECHNIQUES	5.5 LOGIC CIRCUITS5.1
ELECTRONIC INSTRUMENT SYSTEMS	Logic Gates
Revision Log iii	NOT Gate
Measurement Standards iv	Buffer Gate
Basic Knowledge Requirements	AND Gate
Part 66 Basic Knowledge Requirements vi	OR Gate
Table of Contents ix	NAND Gate
	NOR Gate
5.1 ELECTRONIC INSTRUMENT SYSTEMS1.1	Exclusive OR Gate
Typical Arrangement and Layout of	Exclusive NOR Gate
Electronic Instrument Systems 1.1	Negative Logic Gates
Analog Instruments	Digital Circuits
Digital Instruments	Aircraft Logic Circuit Applications
Electronic Displays	Submodule 5 Practice Questions
Electronic Flight Instrument System	Submodule 5 Practice Answers
Engine Indication and Crew Alerting System 1.5	
Submodule 1 Practice Questions	5.6 BASIC COMPUTER STRUCTURE6.1
Submodule 1 Practice Answers	Computer Architecture
	Bits, Bytes, And Words 6.1
5.2 NUMBERING SYSTEMS2.1	Software
Decimal	Hardware
Binary	Central Processing Unit
Binary Place Values	Memory (RAM, ROM, PROM) 6.4
Octal	Integrated Circuits
Hexadecimal	Aircraft Computer Systems
Binary Number System Conversion	Submodule 6 Practice Questions
Binary Coded Decimals	Submodule 6 Practice Answers 6.8
Octal Number System Conversion	
Hexadecimal Number System Conversion 2.4	5.7 MICROPROCESSORS
Submodule 2 Practice Questions	This submodule is only required for B2 Avionics Licensing 7.1
Submodule 2 Practice Answers	
	5.8 INTEGRATED CIRCUITS8.1
5.3 DATA CONVERSION3.1	This submodule is only required for B2 Avionics Licensing 8.1
Analog Data & Digital Data	
Analog Data	5.9 MULTIPLEXING9.1
Digital Data	This submodule is only required for B2 Avionics Licensing 9.1
Data Conversion	
Analog to Digital Conversion	5.10 FIBER OPTICS10.1
Digital to Analog Conversion	Advantages and Disadvantages
Limitations of Conversion	Fiber Optic Data Bus
Submodule 3 Practice Questions	Related Terms
Submodule 3 Practice Answers	Terminations
	Couplers and Terminals
5.4 DATA BUSES4.1	Applications in Aircraft Systems
Data Buses	Submodule 10 Practice Questions
MIL-STD-1553B 4.1	Submodule 10 Practice Answers
ARINC 429	
ARINC 629	5.11 ELECTRONIC DISPLAYS11.1
Aircraft Networks/Ethernet	Principles of Operation
ARINC 664 AFDX	Cathode Ray Tubes
IEEE 1394 Firewire	Light Emitting Diodes
Submodule 4 Practice Questions 4.7	Liquid Crystal Displays
Submodule 4 Practice Answers	Active Matrix Liquid Crystal Display 11.4
	Submodule 11 Practice Questions
	Submodule 11 Practice Answers 11.6



TABLE OF CONTENTS

5.12 ELECTROSTATIC-SENSITIVE DEVICES	12.1
Risks and Possible Damage	12.1
Anti-Static Protection	
Controlled Environment	
Static-Safe Workstation	
Anti-Static Wrist Straps	
Grounding Test Stations	
Ionizers	
Special Handling	
Submodule 12 Practice Questions	
Submodule 12 Practice Answers	
T 40 COSTINADE MANAGEMENT CONTROL	40.4
5.13 SOFTWARE MANAGEMENT CONTROL	
Restrictions And Catastrophic Effects	
Airworthiness Requirements	
Submodule 13 Practice Questions	
Submodule 13 Practice Answers	13.4
5.14 ELECTROMAGNETIC ENVIRONMENT	14.1
Electromagnetic Compatibility (EMC)	14.1
Electromagnetic Interference (EMI)	
High-Intensity Radiated Field (HIRF)	
Lightning/Lightning Protection	
Submodule 14 Practice Questions	
Submodule 14 Practice Answers	
5.15 TYPICAL ELECTRONIC/DIGITAL AIRCRAFT SYSTEMS	
Section A	
Introduction	
Communications, Navigation and Surveillance Systems	15.2
Aircraft Communication Addressing and Reporting System	
(ACARS)	
Flight Control Systems	
Mechanical Flight Control Systems	15.2
	15.2 15.2
Fly by Wire	15.2 15.2 15.3
Fly by Wire	15.2 15.2 15.3
	15.2 15.2 15.3 15.3
Flight Management System (Fms)	15.2 15.3 15.3 15.4
Flight Management System (Fms)	15.2 15.3 15.3 15.4 15.5
Flight Management System (Fms)	15.2 15.3 15.3 15.4 15.5 15.5
Flight Management System (Fms)	15.2 15.3 15.3 15.4 15.5 15.5
Flight Management System (Fms)	15.2 15.3 15.3 15.4 15.5 15.5 15.5
Flight Management System (Fms) Inertial Reference System (IRS) Section B Electronic Flight and Engine Instruments (Ecam/Eicas/Efis) Global Positioning System (Gps/Gnss) Traffic Alert and Collision Avoidance System (Tcas) Integrated Modular Avionics	15.2 15.3 15.3 15.4 15.5 15.5 15.5 15.6 15.7
Flight Management System (Fms) Inertial Reference System (IRS) Section B Electronic Flight and Engine Instruments (Ecam/Eicas/Efis) Global Positioning System (Gps/Gnss) Traffic Alert and Collision Avoidance System (Tcas) Integrated Modular Avionics Cabin Systems	15.2 15.3 15.3 15.4 15.5 15.5 15.5 15.6 15.7
Flight Management System (Fms) Inertial Reference System (IRS) Section B Electronic Flight and Engine Instruments (Ecam/Eicas/Efis) Global Positioning System (Gps/Gnss) Traffic Alert and Collision Avoidance System (Tcas) Integrated Modular Avionics Cabin Systems Information Systems	15.2 15.3 15.3 15.4 15.5 15.5 15.5 15.6 15.7 15.7
Flight Management System (Fms) Inertial Reference System (IRS) Section B Electronic Flight and Engine Instruments (Ecam/Eicas/Efis) Global Positioning System (Gps/Gnss) Traffic Alert and Collision Avoidance System (Tcas) Integrated Modular Avionics. Cabin Systems Information Systems Maintenance Information System	15.2 15.3 15.3 15.4 15.5 15.5 15.5 15.7 15.7 15.8 15.8
Flight Management System (Fms) Inertial Reference System (IRS) Section B Electronic Flight and Engine Instruments (Ecam/Eicas/Efis) Global Positioning System (Gps/Gnss) Traffic Alert and Collision Avoidance System (Tcas) Integrated Modular Avionics Cabin Systems Information Systems	15.2 15.3 15.3 15.4 15.5 15.5 15.5 15.7 15.7 15.8 15.8
Flight Management System (Fms) Inertial Reference System (IRS) Section B Electronic Flight and Engine Instruments (Ecam/Eicas/Efis) Global Positioning System (Gps/Gnss) Traffic Alert and Collision Avoidance System (Tcas) Integrated Modular Avionics Cabin Systems Information Systems Maintenance Information System Submodule 15 Practice Questions	15.2 15.3 15.3 15.4 15.5 15.5 15.5 15.7 15.7 15.8 15.8





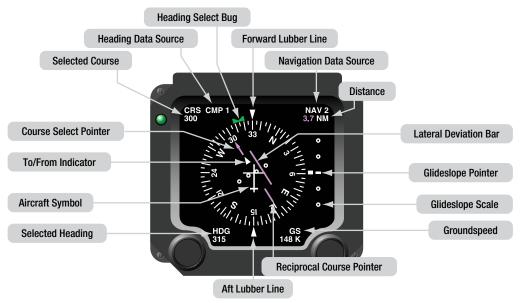


Figure 1-4. Typical EADI (top) and EHSI (bottom) Display Symbology.

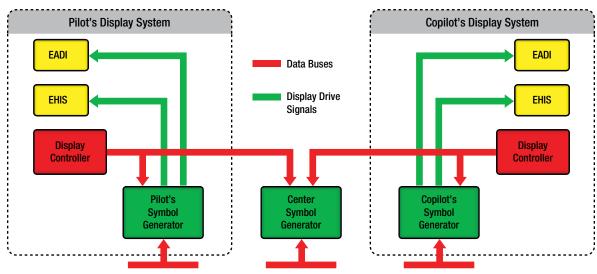


Figure 1-5. Electronic displays are driven by symbol generators.





Figure 1-6. Boeing 777 Electronic Instrument System has 6 LCD Displays.

increase the crew's situational awareness by integrating all of this information into a single composite display instead of the crew having to monitor several independent analog instruments. Also, the colors on the display change to alert the crew to potentially hazardous flight conditions, such as low airspeed, high rate of descent, etc.

Figure 1-7 is a typical Primary Flight Display format showing the artificial horizon in the center of the display, airspeed on the left side, altitude on the right side, heading on the bottom, and flight modes on the top of the display. Notice how the moving ladder format used for altitude and airspeed provide both absolute and relative information so the crew knows not only the exact numeric value, but also the rate that the altitude and airspeed is changing.

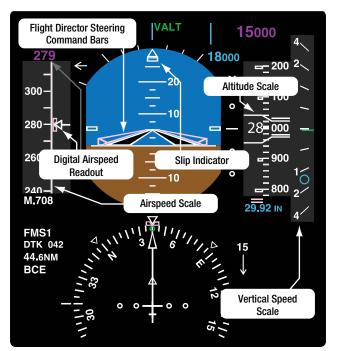


Figure 1-7. Primary Flight Display format.

The Navigation Display, shown in **Figure 1-8**, takes the place of the EHSI display to show the requisite information to navigate the aircraft, including heading, VOR, GPS, and ILS guidance. The ND has the ability to overlay additional information on the navigation page to eliminate the need for separate dedicated

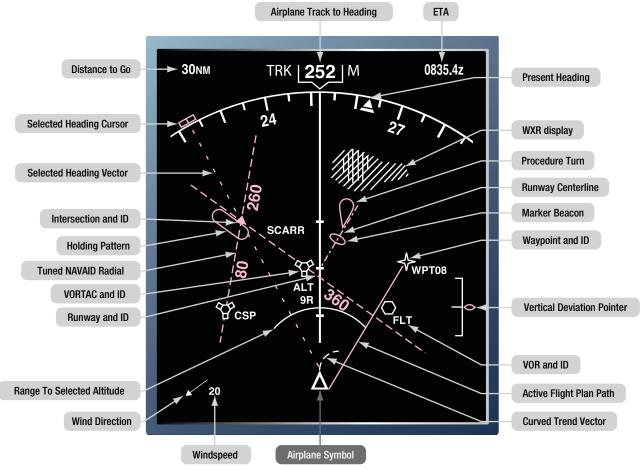


Figure 1-8. Navigation map display format.





Figure 1-9. EICAS engine display format.

displays. Some examples of information that is typically overlaid on the ND include weather information from either the onboard

weather radar (WXR) or ground based sensors, and digital maps showing pre-programmed routes and waypoints from the Flight Management System.

ENGINE INDICATION AND CREW ALERTING SYSTEM

The Boeing Engine Indication and Crew Alerting System (EICAS), also called an Electronic Centralized Aircraft Monitor (ECAM) on Airbus aircraft, performs the monitoring of aircraft systems that was previously performed by the Flight Engineers in 1997. systems that was previously performed by the Flight Engineer in three crew member cockpits. As previously shown in Figure 1-6, the two EICAS displays on the B777 are located in the center instrument panel. The upper EICAS display shows engine performance data, such as pressure ratio, N1 rotor speed, exhaust gas temperature, total air temperature, thrust mode, etc., in addition to cabin pressure, flat/slat position, landing gear position, and crew status alerts. [Figure 1-9]

The EICAS engine display formats mimics the round analog instruments, while also providing digital readouts of the parameters. EICAS improves situational awareness by allowing

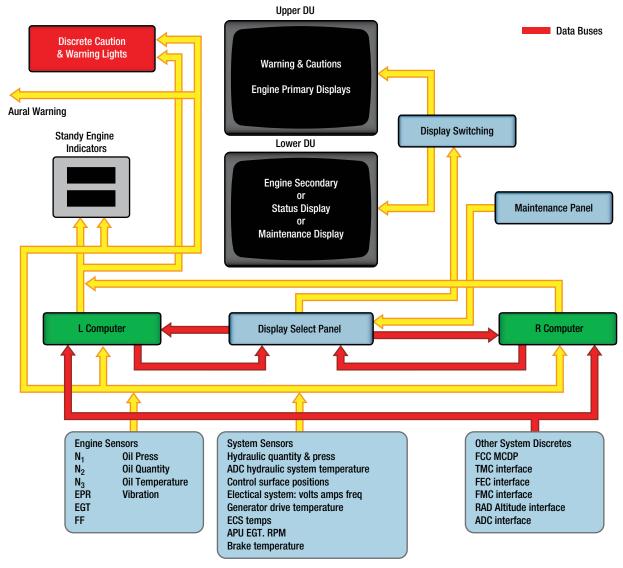


Figure 1-10. EICAS schematic diagram.



the crew to see systems operation in graphical format and alerting them to any failures or impending failures. For example, if low oil pressure is detected, the EICAS will provide an aural alert and show to the oil pressure page on a lower display with a red box outlining which engine has low oil pressure.

The Airbus ECAM system provides the crew with the following levels of warning along with detailed messages as to the nature of the problem and suggested courses of action.

- *Level 3*—An overspeed, fire, or stall condition will cause a repetitive chime aural warning with a bright red flashing light.
- Level 2—A system failure, but not a safety of flight issue, will result in a single chime aural warning and a steady amber light.
- Level 1—Failure leading to system degradation results only in an amber light.
- *Mode or System Status*—If everything is normal, a green light will illuminate.

The lower EICAS display is called a Multi-Function Display because it provides auxiliary information to the flight crew and maintenance crew. The MFD can be used as a secondary engine display, status display, communications display, maintenance page, or electronic checklist. The MFD formats also include synoptic displays that provide system status diagrams for the fuel, electrical, hydraulic, flight control, and environmental control systems, in addition to showing door and landing gear positions. On some aircraft, the MFD is also used to display images from the ground maneuvering camera system.

Figure 1-10 is a schematic diagram of an Engine Indication and Crew Alerting System (EICAS) with all its associated components. The display select panel allows the crew to choose which computer is actively supplying information. It also controls the display of secondary engine information and system status displays on the lower monitor.

EICAS has a unique feature that automatically records the parameters of a failure event to be regarded afterwards by maintenance personnel. Pilots that suspect a problem may be occurring during flight can press the event record button on the display select panel. This also records the parameters for that flight period to be studied later by maintenance. Hydraulic, electrical, environmental, performance, and Auxiliary Power Unit (APU) data are examples of what may be recorded. EICAS uses Built-In-Test Equipment (BITE) for systems and components.

A maintenance control panel is included for technicians. When the aircraft is on the ground, push-button switches display information pertinent to various systems for analysis. [Figure 1-11]

This *Submodule* contained an overview of a state-of-the-art aircraft cockpit with its Electronic Instrument System. The following *Submodules* will discuss how digital data streams are formed and processed by aircraft computers and then sent over digital data buses to cockpit displays to provide essential information for the flight crew and maintenance crew.

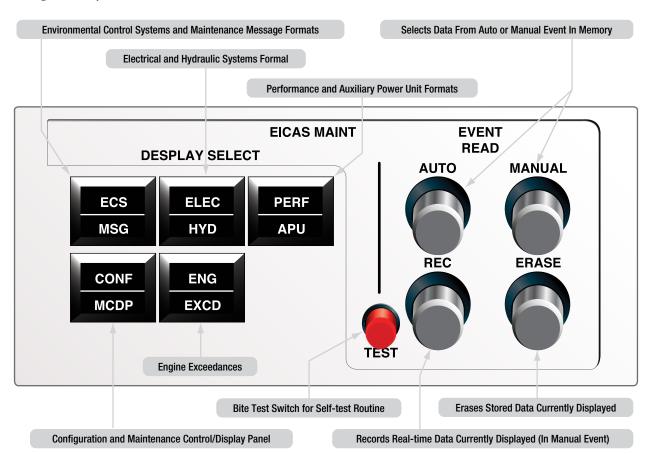


Figure 1-11. EICAS maintenance control panel.



SUBMODULE 1 PRACTICE QUESTIONS

Question 1-1

What are the differences between analog and digital instruments?

Question 1-2

What are the advantages of Liquid Crystal Displays (LCD) over Cathode Ray Tube (CRT) instruments used in "Glass Cockpits"?

Question 1-3

What types of information does an Electronic Attitude Direction Indicator (EADI) and an Electronic Horizontal Situation Indicator (EHIS) provide to the flight crew?

Question 1-4

What is the purpose of the Multi-Function Display (MFD)?

Question 1-5

What information does a Primary Flight Display (PFD) provide?

Question 1-6

What type of information is typically overlaid on the Navigation Display (ND)?

Question 1-7

How information does the Engine Indication and Crew Alerting System (EICAS) provide to improve crew situational awareness?

Question 1-8

What occurs when the flight crew pushes the "event" button on the EICAS Display Select Panel?

